

$$3m0.06=100/500 \times 0.3 = 2/P_1 (P_1 = V_2 V \\ \text{st law } Q-W = \Delta U \\ \text{for isothermal process then } 0 \text{ But } \Delta U = \\ Q=W \\ J = Q = 48283) = -100/500 \ln(0.3 \times 1000 \times 100) = 2/P_1 \ln(P_1 V_1 W = P \\ \text{adiabatic 2-3 process}$$

$$V_3^{\gamma} = P_2 V^{\gamma} P_2 \\ x \left| \frac{1000}{V_3} \right| = \left| \frac{P_3}{P_2} \right| = P_3 \text{ then } P_3^{\gamma} / 0.06 / \frac{2N/m}{0.3} = 52530 = \\ = 0.06 \times 500000 - 0.3 \times 52530 \frac{(1-\gamma)}{1-\gamma} = P_3 V_3 W = (P_3 J 35602$$

(adabatic) $0Q =$

3-1 Process
0W =

$$-\frac{K 153.91}{V} \frac{1}{T_3} = 1^{\gamma} - 0.06 / 0.3 / (293 \text{ then } T = 1^{\gamma-3}) = \left(\frac{V_2}{V_3} \frac{T}{T_2} \right)^{\gamma} \\ J 35603 = 293 - 153.91 (714 \times 0.358) = 3-T1 = mCv (T31Q \\ 0 = 31 W \\ \text{Is } \sum Q = \sum W \\ 31 + W_{23} + W_{12} = W_{31} + Q_{23} + Q_{12} \\ 0 + 35602 + 48283 - = 35603 + 0 + 48283 - \\ \text{st law of thermodynamics is verified! Then the}$$

. The molecular weight , adiabatic index, characteristic gas constant3Ch density and partial pressure of gas mixture‘

This gas is N_2 10%, O_2 60%, H_2 30% EX. The volumetric analysis of a gas is as follows: N volume of air. Take the molar specific heat 1 volume of gas to 2 mixed with air in proportions oxygen by 21% kJ/kg mol K . Air contains 20 capacity at constant volume for diatomic gas = bar for 1.3 °C and 30 .If the air-gas mixture is kept at V_2 volume and the remainder is N : temperature and pressure respectively. Determine for the air-gas mixture

- (a) The mean relative molecular mass of the mixture (M_{av})
- (b) The value of the adiabatic index of the mixture (γ)
- (c) The characteristic gas constant for the mixture (R)
- (d) The density of air
- (e) in the mixture, The partial pressure and mass fraction of O_2

$$\text{kg/kg.mol} 18.146 = \frac{32 \times 0.21 + 28 \times 0.79 \times 1 + 32 \times 0.1 + 2 \times 0.6 + 28 \times 0.3 \times 2}{3} M_{av} = 3Q$$

$$\bar{C}_{vav} = \frac{\bar{k}_J}{\sum n} = 20 = 20 \lambda x^{0.21 + 0.79 x_1 + 0.1 + 0.6 + 0.3 x_2} = \sum n C_v = \frac{\sum n C_v}{3}$$

$$Rm = \bar{C}_p - \bar{C}_v$$

$$-\frac{28.3143}{p_{av}} = 20 + 8.3143 = C \frac{k_J}{kg.mol}$$

$$-\frac{1.4157}{\bar{C}_{vav}} = \frac{28.3143}{20} \gamma = \frac{C_{pav}}{20} = K$$

$$R_{av} = \frac{R_m}{M_{av}} = \frac{0.4582}{18.146 M} = \frac{8.3143}{kg.}$$

$$air = \frac{P_t}{n_T} \frac{R}{M_{air}} = \frac{KN}{3} = 13.334 = 1 x^{130} \frac{K}{^2 m}$$

$$air R \frac{R}{M_{air}^m} = \frac{8.3143}{28.84(28x79+.32x0.21)} = \frac{0.28829}{K} = \frac{8.3143}{kg.}$$

$$air \rho \frac{P_{air}}{(30+273)288.29 R} = \frac{1000x13.334}{T} = \frac{0.496}{^3 m}$$

$$_{2o} = \frac{P_t}{n_T} \frac{R}{M_{2o}} = \frac{KN}{3} = 17.766 = 0.21 + 0.1 x 2 (130) =$$

$$kg 54.441 = 18.147 x 3 m_T = n_T M_{av} =$$

$$kg 13.12 = 0.21 + 0.1 x 2 (32) = m_o$$

$$24.1\% = \frac{m_o}{m_T} \%$$